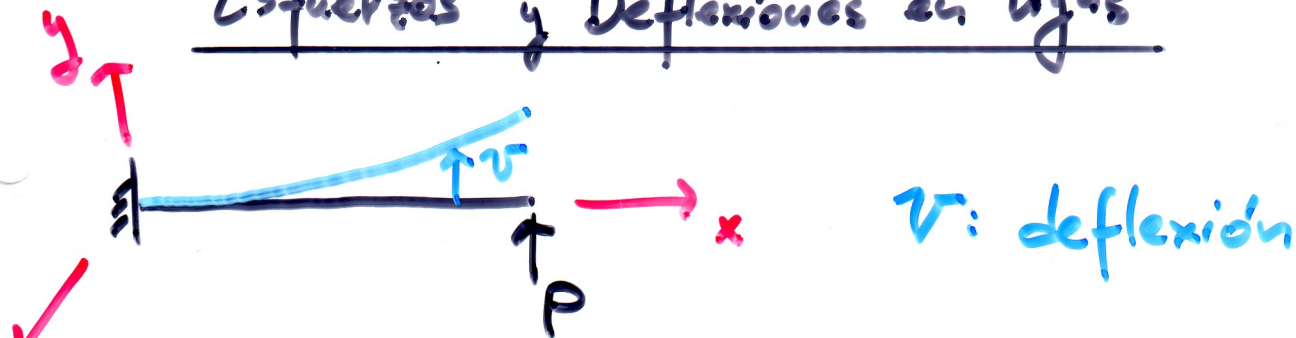


Esfuerzos y Deflexiones en vigas



v : deflexión

Consideraciones:

La Sección transversal simétrica con respecto al eje y .

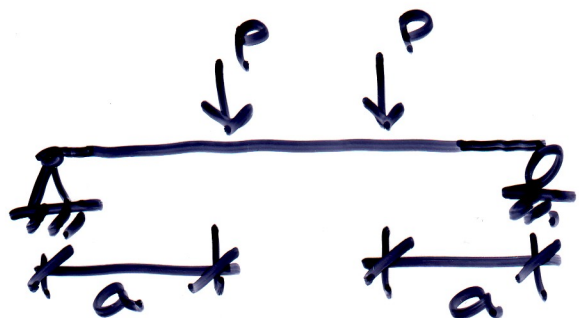
La Flexión ocurre en el mismo plano vertical

Flexión Pura



$$V(x) = 0$$

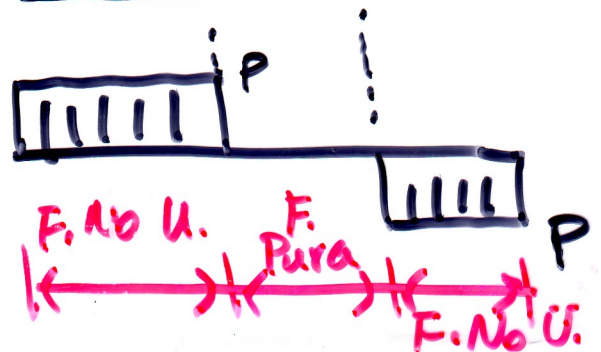
Flexión No Uniforme



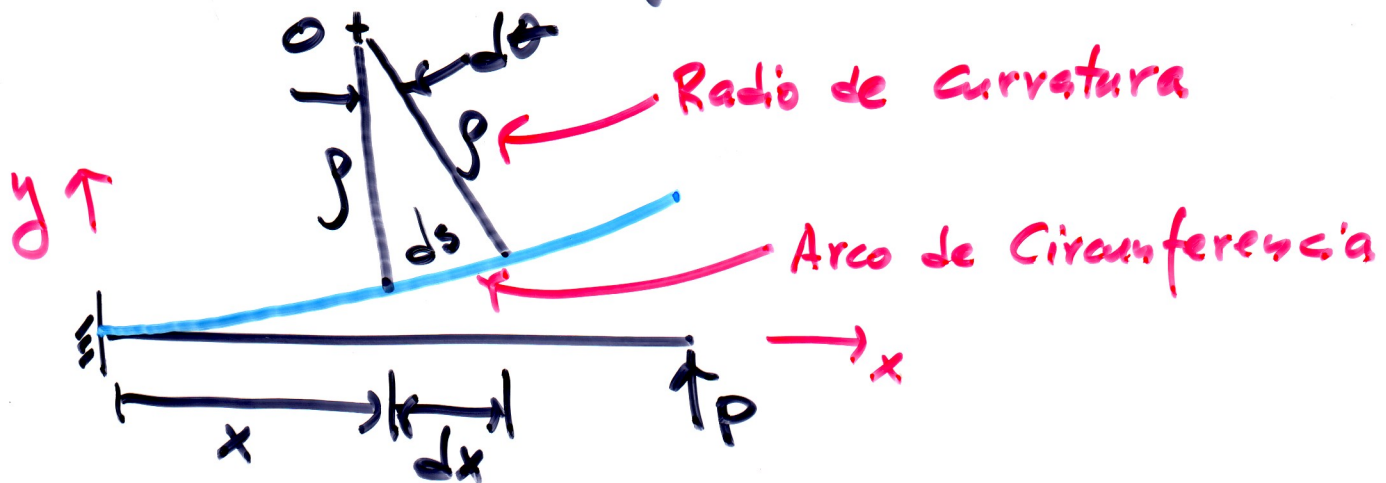
$M(x)$



$V(x)$



Curvatura de una viga



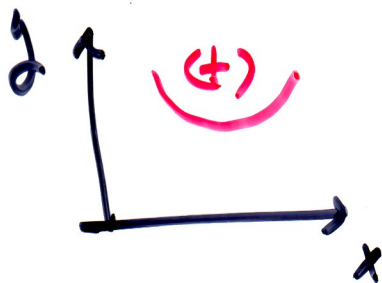
$$\text{Curvatura} \sim \kappa = \frac{1}{\rho}$$

$$\text{Arco de Circunferencia} \sim ds = \rho d\theta$$

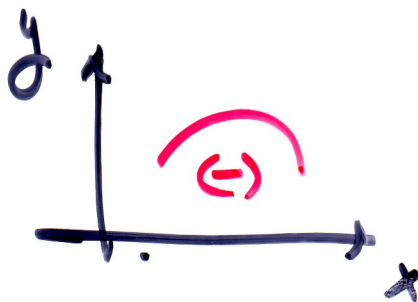
$$\kappa = \frac{1}{\rho} = \frac{d\theta}{ds}$$

Aplicando la teoría de deformaciones pequeñas (la curva deflectada es casi plana): $ds \approx dx$

$$\Rightarrow \kappa = \frac{1}{\rho} = \frac{d\theta}{dx}$$



Curvatura $(+)$

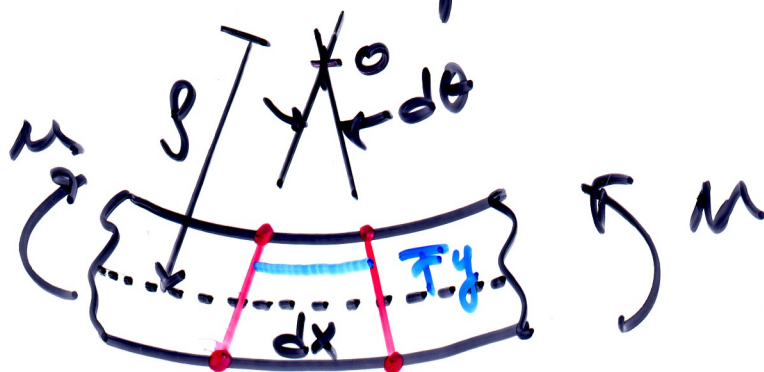
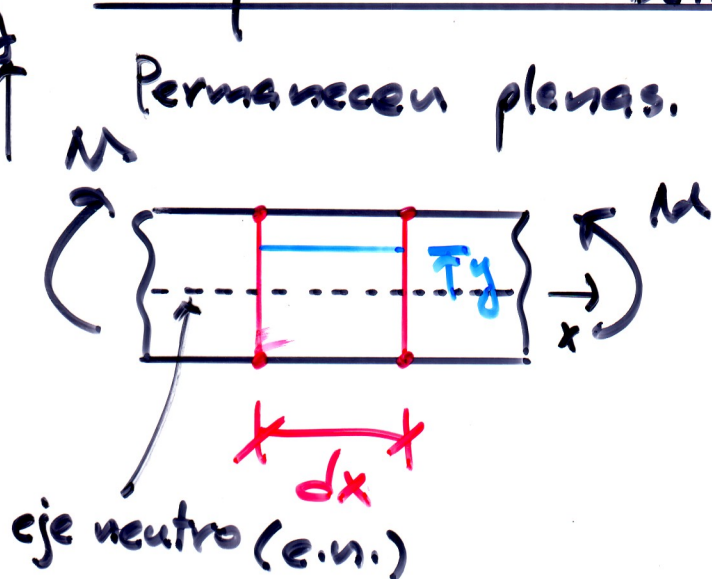


Curvatura $(-)$

Deformación Unitaria Longitudinal

Principio de Navier-Bernoulli: Secciones planas

Permanecen planas.



Deformación unitaria de un punto a una distancia " y " del eje neutro:

$$\epsilon_x = \frac{\Delta L}{L_0} = \frac{L_f - L_0}{L_0}$$

$$\hookrightarrow L_f = (\rho - y)d\theta = dx - yd\theta = dx - y \frac{dx}{\rho}$$

$$\hookrightarrow L_0 = dx$$

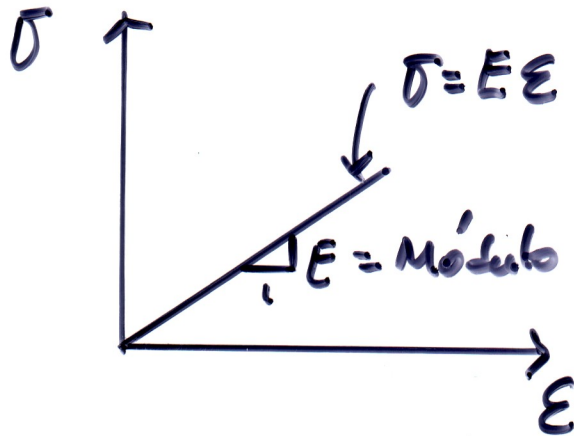
$$\Rightarrow \epsilon_x = -\frac{y}{\rho} = -ky$$

$\hookrightarrow \epsilon_x^{(-)} \rightarrow$ compresión

$\epsilon_x^{(+)} \rightarrow$ tensión

Esfuerzos Normales en Vigas

Ley Constitutiva: Ley que relaciona σ vs ϵ . En este curso se utilizará la Ley de Hooke $\leadsto \sigma = E\epsilon$



$\hookrightarrow$ Materiales Linealmente Elásticos.

$$\sigma_x = E\epsilon_x = -E \frac{y}{\rho} = -KEy$$

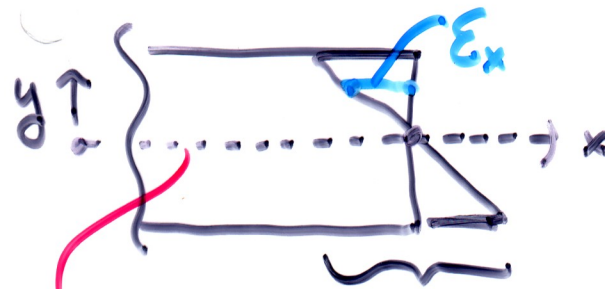


Diagrama de Deformaciones Unitarias

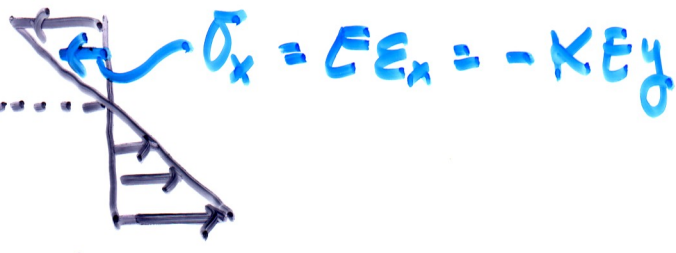
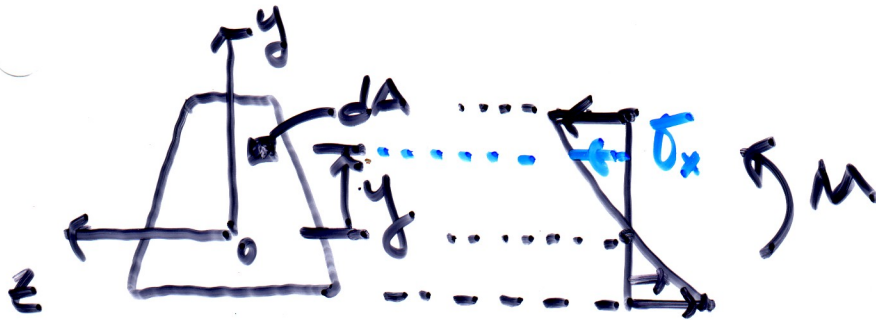


Diagrama de Esfuerzos

$\sigma_x (+)$ $\leadsto$ Tensión

$\sigma_x (-)$ $\leadsto$ Compresión

Ubicación del Eje Neutro



$$\sum F_x = 0$$

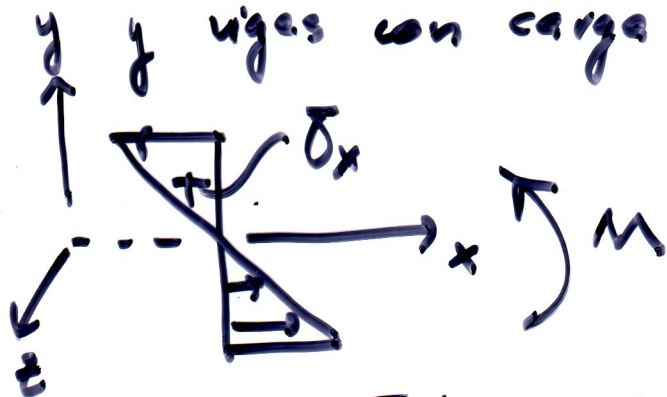
$$\int_A \sigma_x dA = - \int_A KEy dA = 0$$

$$\hookrightarrow \int_A y dA = 0$$

=0 El eje neutro está localizado en el centroide de la sección transversal.

↳ Válido para materiales linealmente elásticos

y vigas con carga axial nula.



$$dM = \overbrace{(-\sigma_x dA)}^{dF_x} y$$

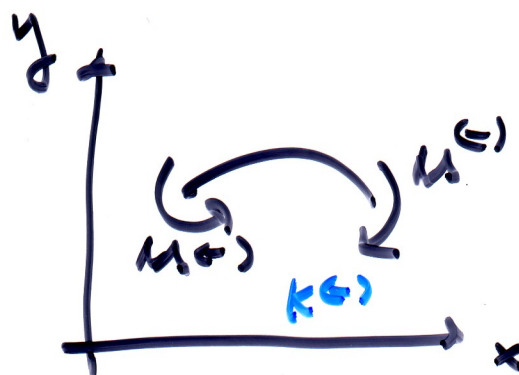
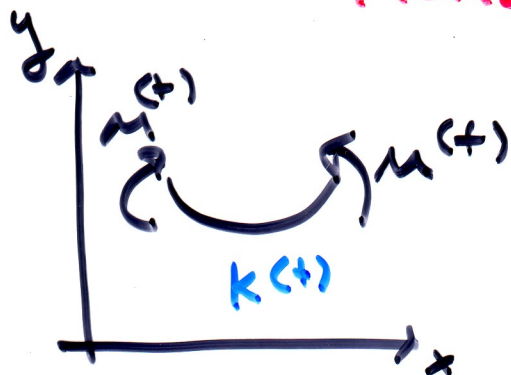
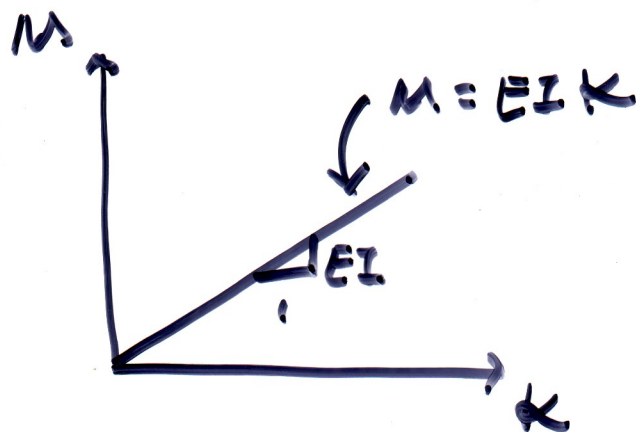
$$\sum M_z = M = - \int_A \sigma_x y dA$$

$$M = \int_A KEy^2 dA = KE \int_A y^2 dA$$

$$\Rightarrow M = KEI$$

$$\Rightarrow k = \frac{1}{\rho} = \frac{M}{EI}$$

Rigidez
Flexionante

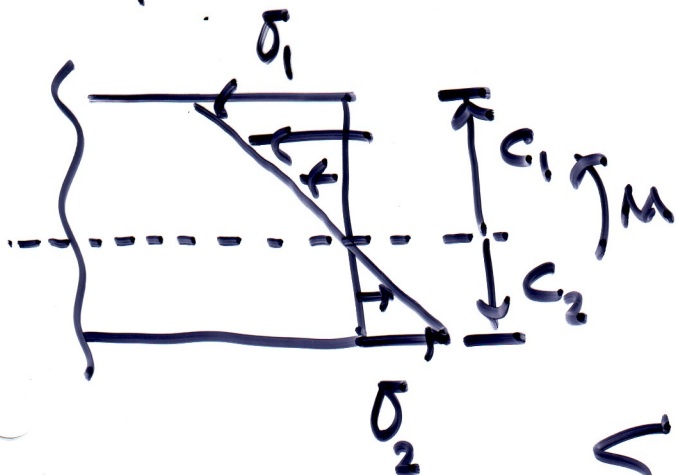


Calculo de Esfuerzos Normales:

$$\sigma_x = -k E y = -\frac{M}{EI} E y \leadsto \boxed{\sigma_x = -\frac{M y}{I}}$$

Esfuerzos de Flexión

Esfuerzos Máximos:

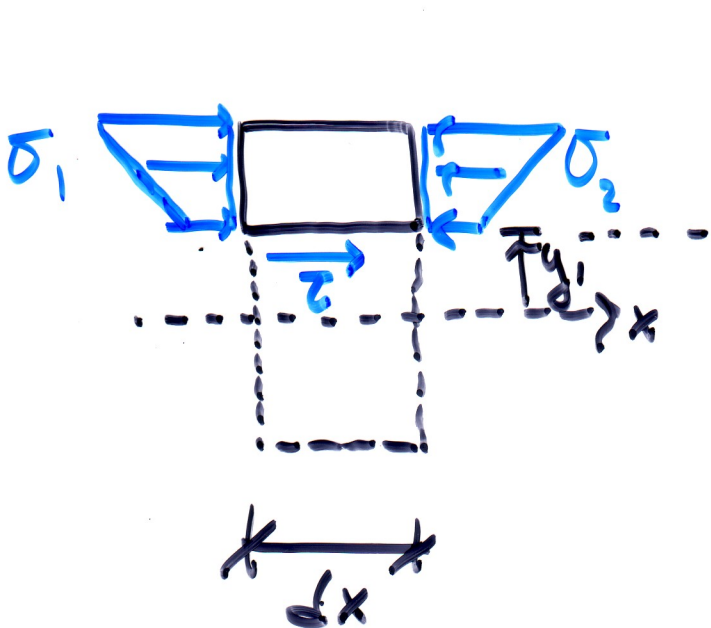
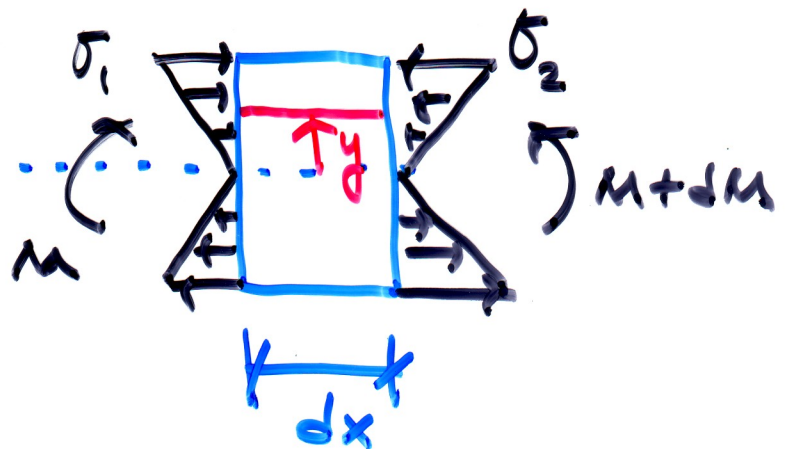
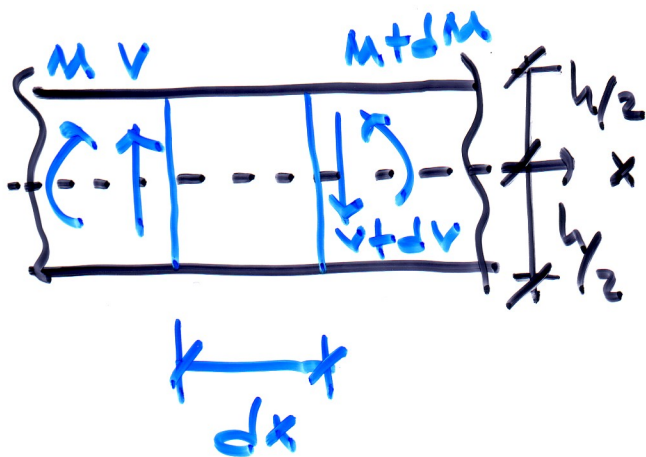
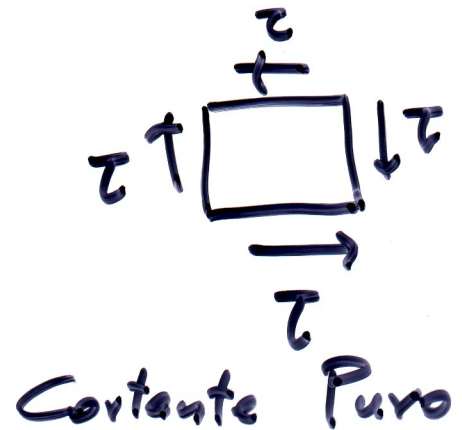
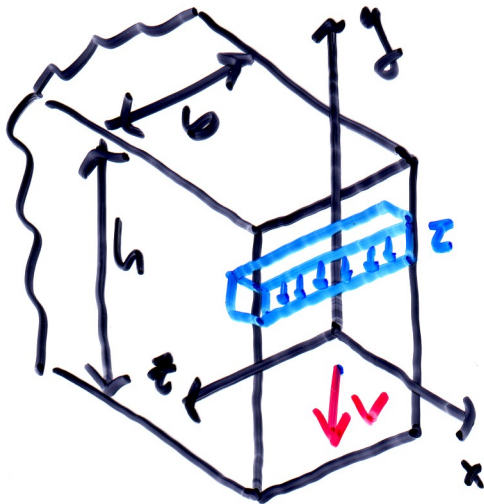


$$\sigma_1 = -\frac{M c_1}{I} ; \sigma_2 = \frac{M c_2}{I}$$

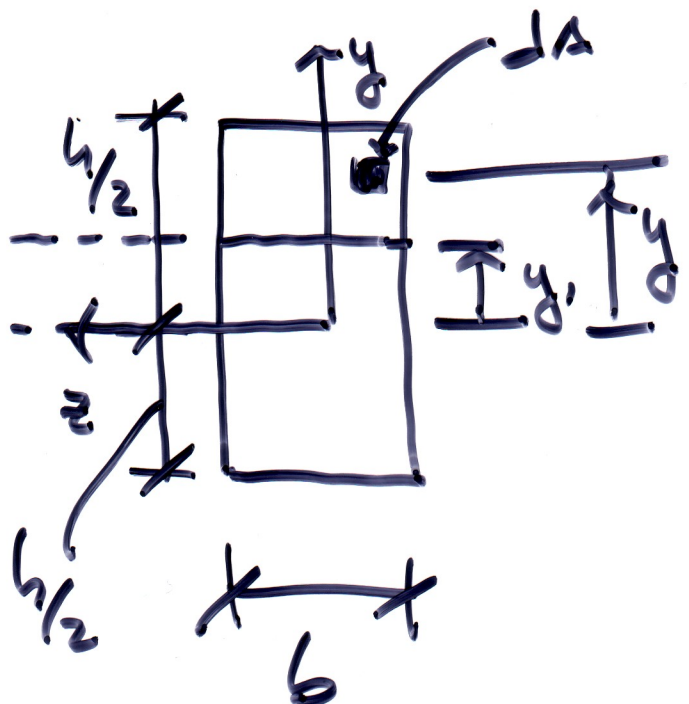
$$S_1 = \frac{I}{c_1} ; S_2 = \frac{I}{c_2}$$

S : Módulo de Sección

Esfuerzos Cortantes en Vigas Rectangulares



Sección Longitudinal



Sección Transversal

$$\sigma_1 = \frac{My}{I}$$

;

$$\sigma_2 = \frac{(M+dM)y}{I}$$

$$\sum F_x = 0$$

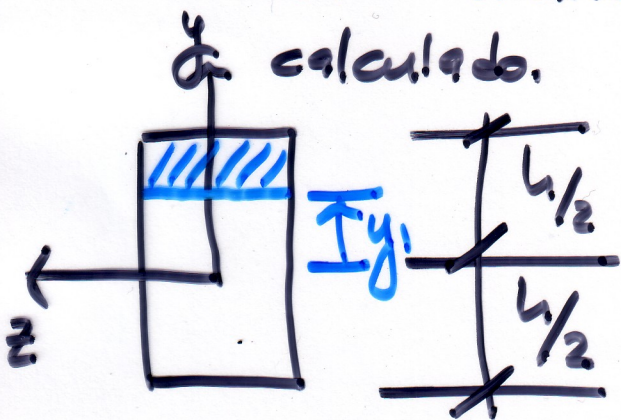
$$\int \sigma_1 dA + \tau b dx - \int \sigma_2 dA = 0$$

$$\int \frac{My}{I} dA - \int \frac{(M+dM)y}{I} dA + \tau b dx = 0$$

$$\tau = \frac{dM}{dx} \left(\frac{1}{Ib} \right) \int y dA \rightsquigarrow$$

$$\tau = \frac{VQ}{Ib}$$

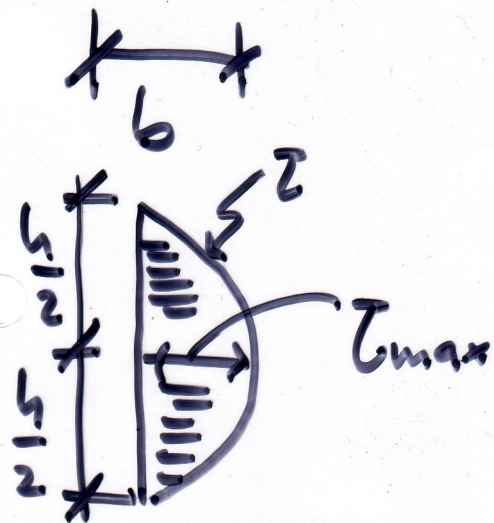
Q : Primer momento del área con respecto al eje neutro al nivel al cual el esfuerzo cortante es calculado.



$$Q = b \left(\frac{h}{2} - y_1 \right) \left[\frac{1}{2} \left(y_1 + \frac{h}{2} \right) \right]$$

$$Q = \frac{b}{2} \left(\frac{h^2}{4} - y_1^2 \right)$$

$$\Rightarrow \tau = \frac{V}{2I} \left(\frac{h^2}{4} - y_1^2 \right)$$



$$\tau_{max} = \frac{Vh^2}{8I} = \frac{3}{2} \frac{V}{bh}$$